

Free rigid body rotation matrix

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Let \mathbf{i} , \mathbf{j} and \mathbf{k} denote the unit vectors, directed along the principal axes of inertia of a rigid body, corresponding to the principal momenta A , B and C . The three coordinate functions p , q and r of the angular velocity $\boldsymbol{\omega} = p\mathbf{i} + q\mathbf{j} + r\mathbf{k}$ of such freely rotating rigid body are then determined for a given (conserved) angular momentum \mathbf{m} and (twice the) energy $h := \boldsymbol{\omega} \cdot \mathbf{m} = Ap^2 + Bq^2 + Cr^2$.

Each row of the (orthonormal) matrix

$$S := \begin{pmatrix} \alpha_0 \cos \psi - \alpha_1 \sin \psi & \alpha_0 \sin \psi + \alpha_1 \cos \psi & Ap/m \\ \beta_0 \cos \psi - \beta_1 \sin \psi & \beta_0 \sin \psi + \beta_1 \cos \psi & Bq/m \\ \gamma_0 \cos \psi - \gamma_1 \sin \psi & \gamma_0 \sin \psi + \gamma_1 \cos \psi & Cr/m \end{pmatrix},$$
$$\begin{pmatrix} \alpha_0 \\ \beta_0 \\ \gamma_0 \end{pmatrix} := \frac{1}{\sqrt{\omega^2 - h^2/m^2}} \begin{pmatrix} (1 - Ah/m^2)p \\ (1 - Bh/m^2)q \\ (1 - Ch/m^2)r \end{pmatrix}, \quad \begin{pmatrix} \alpha_1 \\ \beta_1 \\ \gamma_1 \end{pmatrix} := \frac{1}{\sqrt{m^2\omega^2 - h^2}} \begin{pmatrix} (B - C)qr \\ (C - A)rp \\ (A - B)pq \end{pmatrix},$$
$$\psi = \psi(t) := \frac{ht}{m} + \left(\frac{h}{m} - \frac{m}{A}\right) \left(\frac{h}{m} - \frac{m}{B}\right) \left(\frac{h}{m} - \frac{m}{C}\right) \int_0^t \frac{dt}{\omega^2 - h^2/m^2},^1$$

which we calculated in [1], provides the three coordinates of a corresponding rotating unit vector with respect to an inertial (that is, a non-rotating) frame which directing unit vectors shall be denoted as \mathbf{x} , \mathbf{y} and $\mathbf{z} := \mathbf{m}/m$.²

Accordingly, each column of the matrix S provides the three coordinates of a corresponding non-rotating unit vector with respect to the rotating frame. Thus, the matrix S might formally be construed via the relation

$$\begin{pmatrix} \mathbf{i} \\ \mathbf{j} \\ \mathbf{k} \end{pmatrix} = S \begin{pmatrix} \mathbf{x} \\ \mathbf{y} \\ \mathbf{z} \end{pmatrix}.$$

We might also readily express the angular velocity $\boldsymbol{\omega}$, with respect to the non-rotating frame, as $\boldsymbol{\omega} = \sqrt{\omega^2 - h^2/m^2} (\cos \psi \mathbf{x} + \sin \psi \mathbf{y}) + h/m \mathbf{z}$.

Acknowledgment

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References

- [1] Adlaj S. *Torque free motion of a rigid body: from Feynman wobbling plate to Dzhanibekov flipping wingnut*. Available at <https://semjonadlaj.com/SP/TFRBM.pdf>.
- [2] Kopytov N.P. *Using wireless sensors to monitor rotational motion of a "free flight" of a rigid body*. July 30, 2024 Session of the V.V. Shevchenko Seminar on Algebraic Methods in Theoretical Mechanics. Available at https://www.mathnet.ru/php/seminars.phtml?presentid=43777&option_lang=eng.

¹We have used the non-boldface letters ω and m to denote the moduli of the angular velocity and the angular momentum, respectively, so $\omega^2 = \boldsymbol{\omega} \cdot \boldsymbol{\omega} = p^2 + q^2 + r^2$ and $m^2 = \mathbf{m} \cdot \mathbf{m} = A^2p^2 + B^2q^2 + C^2r^2$.

²We have tacitly excluded the case where $\boldsymbol{\omega}$ and \mathbf{m} are colinear, that is, the case of the vanishing difference $\omega^2 - h^2/m^2$, including the critical motion of the Dzhanibekov top which was addressed in [1]. Note that the condition $(Ah - m^2)(Bh - m^2)(Ch - m^2) = 0$ implies that ψ is linear (as a function of time) although it does not imply the constancy of the angular speed ω , whereas the condition $(A - B)(B - C)(C - A) = 0$ does imply the constancy of ω along with the linearity of ψ .